SPECIAL AIRWORTHINESS INFORMATION BULLETIN



U.S. Department of Transportation Federal Aviation Administration

AIRCRAFT CERTIFICATION SERVICE 800 INDEPENDENCE AVENUE, S.W. WASHINGTON, DC 20591

No. ACE-99-10 November 24, 1998

Published by: FAA, AFS-610, P.O. Box 26460, Oklahoma City, OK 73125

This is issued for informational purposes only and any recommendation for corrective action is not mandatory.

Introduction

The purpose of this Special Airworthiness Information Bulletin (SAIB) is to inform registered owners of certain amateur built Van's Aircraft, Inc. (Van's), RV-3/-3A model airplanes of design, equipment and operating limitations that Van's considers critical for safe aerobatic flight. Adherence to Van's specifications is highly recommended.

Background

Eight accidents with seven fatalities have occurred in the United States, since 1980. (One additional fatal accident occurred in Canada.) Five of the nine accidents (including the Canadian accident) involved unmodified RV-3s. Four airplanes were modified to Van's RV-3A configuration (optional designation after incorporating Van's first wing structural modification (CN-1) for Type I wing spars).

RV-3/-3A U.S. Accident Brief Summary (Source: NTSB Accident Briefs; may not include all causal factors leading to the accident):

		NTSB			Fatal		<u>Accident</u>
No.	<u>Date</u>	File No.	'N' Number	Model	Y/N?	Location	<u>Description</u>
1.	8/5/80	3-2810	N66CJ	RV-3	Y	Streator,	Cruise flt.; left wing separated.
						IL	(Acft. rebuilt from prev. landing
							accident.) (Pre-mod. accident.)
2.	3/11/81	3-0469	N55F	RV-3	Y	Kennett,	Low level/high speed flyby; left wing
						MO	separated from acft. (Undersized wing
							spar attach bolts.)
3.	11/27/86	2051	N135RV	RV-3	Y	Carmel	Acro; wings separated at top of climb.
						Valley, CA	
4.	3/14/88	578	N87JP	RV-3A	N	Antioch,	Acro; spin recovery, left wing bent
						CA	upward, impacted nearly level attitude.
5.	8/6/92	2667	N50WP	RV-3	Y	Jackson,	Acro; abrupt pitch up/top of loop,
						MI	wings folded. (CN-1 spar mod.)
6.	10/8/95	1760	N27RV	RV-3A	Y	Forest	Acro; rapid climbing; r. wing
						Grove, OR	separated; +9.2 g's indicated.
7.	3/8/98	FTW98F	N99HV	RV-3A	Y	Elbert, CO	Acro; wing folded top of loop.
		A145					
8.	9/26/98	LAX98L	N244WW	RV-3	Y	Arbuckle, CA	VFR only pilot, in 400'overcast
		A305					instrument meteorological conditions
							(IMC); pulled up from dive; +7/-4.5
							g's indicated.

The FAA previously issued two General Aviation Notices (GENOTs) addressing RV-3 inflight wing failures:

- a. GENOT #8600.14, RWA 1/40 SVC B, dated 3/24/81, provided for the suspension of RV-3 airworthiness certificates until new certificates could be issued prohibiting aerobatic flight, and
- b. GENOT #8600.25, RWA 2/100 SVC B, dated 5/22/82, lifted the aerobatic restrictions based on a modification of the wing structure and the individual airplane's demonstrated aerobatic flight up to design limit loads.

Van's Design, Equipment and Operating Recommendations

Van's recommends that <u>aerobatic flight operations be prohibited</u> for RV-3/-3A airplanes that have not had Van's wing spar modifications incorporated, or those modified RV-3/-3A airplanes that are over 1,050 pounds gross weight (including pilot and wing fuel weight).

- a. <u>Design and Weight Limitations</u>: RV-3/-3A airplanes properly modified to Van's Change Notice #1 (CN-1) and Change Notice #2, Type I (CN-2-I) Spar Modification Instructions for <u>Type I wings</u> (1/8 inch thick spar bar stock with 1/8 inch rivets) or CN-2-II Spar Modification Instructions for <u>Type II wings</u> (3/16 inch thick spar bar stock with 3/16 inch rivets) should not be prohibited by design from aerobatics, unless the airplane's aerobatic gross weight (including pilot and wing fuel weight) exceeds 1,050 pounds. (Van's CN-1 (re-enforcement of the root rib and rear spar) if not previously incorporated, is designed to be incorporated in conjunction with CN-2-I.)
- b. <u>Required Aerobatic Equipment and Operating Limitation Placards</u>: In addition to Van's wing spar modifications, Van's RV-3/-3A design includes the following minimum <u>flight equipment and placard installation</u> for aerobatic flight operations:

Placards:

- 1. + 6 G maximum positive load limit at maximum aerobatic gross weight of 1,050 lbs. (including pilot and wing fuel weight).
- 2. -3 G maximum negative load at 1,050 lbs. gross weight (including pilot and wing fuel weight).
- 3. 210 mph never exceed speed (V_{NE}).
- 4. 132 mph design maneuvering speed (V_A).

Equipment

1. Recording accelerometer (G meter) required.

Recommendation

<u>Design</u>, <u>Equipment and Operation</u>: The FAA highly recommends registered owners of amateur built RV-3/-3A model airplanes incorporate Van's required design changes, placards and equipment prior to normal or aerobatic flight and observe Van's aerobatic flight operational limitations. In no case should the airplane be operated above 1050 pounds max. gross weight, above +6/-3 g's, or beyond V_{NE} or V_A .

Overload: In the event of inadvertently encountering inflight "g" loads above +6/-3 g's, it is highly recommended that registered owners immediately contact Van's for a detailed inspection criteria. A detailed wing structural inspection in accordance with Van's design criteria should be completed prior to the next flight. Any structural damage should be corrected prior to the next flight using specifications obtained from Van's.

<u>Condition Inspection</u>: If you are not the original builder, you should not assume your RV-3 meets all of Van's design specifications, unless a thorough internal inspection has been conducted to your satisfaction. When documenting the wing's internal condition, the inspection should include, but not be limited to, the following:

- 1. Evidence of aircraft quality workmanship standards;
- 2. Over/under driven rivets;
- 3. Corrosion;
- 4. Insufficient edge distance for rivets and center section bolts, or oversized/elongated holes;
- 5. Extra holes in spar flange strips;
- 6. Corroded, damaged or defective control push rod linkages;
- 7. Cracks around aileron bellcrank mount in the wing rib web;
- 8. Evidence of previous damage to the internal wing structure, i.e., bent ribs, etc.

FOR FURTHER INFORMATION CONTACT:

FAA, Chicago Aircraft Certification Office, Attention: Mr. Nick Miller, 2300 East Devon Avenue, Des Plaines, IL 60018, telephone (847) 294-7837, facsimile (847) 294-7834. Van's Aircraft, Inc., may be contacted at P.O. Box 160, North Plains, OR 97133, telephone (503) 647-5117, facsimile (503) 647-2206.